

engineering A design





RIFLE SCOPE MOUNT FORCE SIMULATION & DESIGN OPTIMIZATION

TEAM: IRONSIGHT

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BACKGROUND WHAT IS DESIRED?

Client Goals/Deliverables

- Develop an optimized riflescope mount
 - Focus: weight reduction
- Process refinement and delivery
- **Design Constraints**
 - Weight < 3.69 oz
 - ≥ Strength of current design
 - Cost effective







Mises (ksi) 3.326e+01 3.049e+01 2.772e+01 2.494e+01 2.217e+01 1.940e+01 1.663e+01 1.109e+01 8.315e+00 2.772e+00 1.678e-07



OVERVIEW

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Governing Principles

- Mathematical Model ۲
- Proof of Concept ٠

Concept Development

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- Material Redesign •
- Geometry Redesign •





• Live Fire Testing Data Analysis



CONCEPT DEVELOPMENT MATERIAL REDESIGN IN REVIEW

Material Indexing	 Modulus of Elasticity(E) Density (ρ) Yield Stress (σ) Cost per Unit Volume (C_v)
	Material Compatibility - Corrosion Study - ASTM B117 Salt Spray Test - Hard coating - MIL-A 8625 Type III
	Material Pricing• Unit and Bulk Quotation• Purchase for Prototype

	Candidate M	laterial Unit and	Bulk Pricing	Maximum Corrosion rate (mm/year) of treated and untreated alloys in salt spray tests							
	Al 7075	Mg AZ61A	Al 2024	Al 7068		Al-7075	Mg-AZ61A				
Unit	13.82	4.06-42.89	15.35-	19.77			J				
(\$/lb)			23.16		Base Metal	0.08587 (3)	17.96 (1)				
Bulk	1.13	4.06-11.56	N/A	N/A	Hard Coated	0.015 (4)	0.022 (2)				
(\$/ID)					 (1) Dhanapal et al., "Experimental Investigation of the Corrosion Behavior of Friction Stir Welded AZ61A Magnesium Alloy Welds under Salt Spray Corrosion Test and Galvanic Corrosion Test Using Response Surface Methodology." (2) Sreekanth, Rameshbabu, and Venkateswarlu, "Effect of Various Additives on Morphology and Corrosion Behavior of Ceramic Coatings Developed on AZ31 Magnesium Alloy by Plasma Electrolytic Oxidation." 						

(3) Al-Moubaraki and Al-Rushud, "The Red Sea as a Corrosive Environment. (4) P. B et al., "Effect of Anodization on the Corrosion Be





Material Indices for Four Relevant Parameters

Behavior	of Aluminium	Alloy in	HCl Acid	and NaOH	."

	Al 7075	Mg AZ61A	Al 2024	Al 7068
$rac{E^{1/2}}{ ho}$	999.8	<u>1240.2</u>	<u>1076.5</u>	999.67
$rac{\sigma^{2/3}}{ ho}$	171.2	159.5	139.1	<u>188.7</u>
$\frac{E^{1/2}}{C_v}$	72.3	<u>152.1</u>	<u>79.3</u>	69.1
$\frac{\sigma^{2/3}}{C_v}$	12.4	<u>19.57</u>	10.3	<u>13.0</u>

Materials Indexing Results



GOVERNING PRINCIPLES MATHEMATICAL MODEL

Impulse Momentum Theorem..... $F * \Delta t = m * \Delta v$

After rearranging..... $F = m * \frac{\Delta v}{\Delta t}$ Simplification..... F = m * a







GOVERNING PRINCIPLES PROOF OF CONCEPT

 Force from swinging pendulum impact force matched to acceleration in bar



Working principle behind the impact test stand

Impact test stand

DATA COLLECTION LIVE FIRE TESTING

Collection Devices

- Accelerometer Dytran 3313
- Daq Vibration Research
- Custom Accelerometer mount
- FN MK 17 Rifle

Testing

• 80 recorded live fire tests

• Three different shooters

Outcome

• 1.6-1.9 million points of data

(Left) Live fire setup for acceleration data gathering on mount. (Right) Dytran 3313

DATA COLLECTION INITIAL DATA ANALYSIS

X Axis – 726 lbf
Y Axis – 741 lbf
Z Axis – 652 lbf

DESIGN OPTIMIZATION SOLIDWORKS SIMULATION SETUP

Jaws of mount are fixed

Hardware is estimated with rigid connections Scope is rigidly connected to mount base and rings

DESIGN OPTIMIZATION SOLIDWORKS SIMULATION FORCES

Forces are applied to the scope

- X = 726 lbf (Along scope)
- Y = 741 lbf (Across scope)
- Z = 652 lbf (Vertical)

DESIGN OPTIMIZATION FINITE ELEMENT ANALYSIS RESULTS

DATA COLLECTION REVISED DATA ANALYSIS

Accelerations

- Averaged Accelerations
- Strain Rate Sensitivity
- X Axis 39 G
- Y Axis 36 G
- Z Axis 26 G

- X Axis 137 lbf
- Y Axis 126 lbf
- Z Axis 93 lbf

7 lbf 5 lbf lbf

DESIGN OPTIMIZATION REVISED SOLIDWORKS SIMULATION FORCES

Forces are applied to the scope

- X = 137.27 lbf (Along scope)
- Y = 126.04 lbf (Across scope)
- Z = 93.16 lbf (Vertical)

DESIGN OPTIMIZATION HIGH STRESS REGION IDENTIFICATION

Stress plots overlaid on the Ultramount 3D model. Hotter colors represent areas of higher stress

- Forces are applied to the Ultramount to locate high stress areas
 - Between rear scope ring and jaws
 - Infront of front jaw toward front scope ring

High stress areas were noted as points to reinforce for future designs

1.386e+01

von Mises (ksi)

3.326e+01

3.049e+01

2.772e+01

2.494e+01

2.217e+01

1.940e+01

1.663e+01

- 1.109e+01
- 8.315e+00
- 5.543e+00
- 2.772e+00
- 1.586e-07

DESIGN OPTIMIZATION DEVELOPMENT

3 new designs were developed over the course of this project

- Designs 1 and 2 are based on the Ultramount design
- Design 3 is a new design focused on weight reduction

DESIGN OPTIMIZATION DESIGN 1

Mass: 3.541 oz

DESIGN OPTIMIZATION DESIGN 2

Mass: 3.339 oz

DESIGN OPTIMIZATION DESIGN 3

Mass: 2.609 oz

DESIGN OPTIMIZATION FINITE ELEMENT ANALYSIS

The new Designs were placed under the same analysis as the Ultramount for comparison

- Designs 1 and 2 are much stronger ullet
- Design 3 is significantly lighter but \bullet weaker than the original

	Design Sta	ts. Table	
Image	Design #	Mass (oz) / % Reduction	Peak Stre (ksi)*
2 4 2 in-LBS	Original	3.689 / 0%	33.26
	1	3.541 / 4%	21.98
	2	3.339 / 9%	24.07
	3	2.609 / 29%	54.79

VALIDATION 3D PRINTED MODELS

Models were 3D printed to validate our simulation methods

- Weights were hung from the front and back rings of the mounts
- Recorded relative displacements
- Test recreated in Solidworks

3D Printed models

Setup for displacement testing

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3D Printed Prototype Displacement Results

(Disp	placement in inches)	Ultramount	Design 1	Design 2	
ation	Displacement (Back Ring)	0.00584	0.007671	0.007763	
Simul	Displacement (Front Ring)	0.02119	0.02231	0.02721	
nental	Displacement (Back Ring)	0.014	0.01	0.033	
Experin	Displacement (Front Ring)	0.042	0.077	0.09	

Solidworks simulation for comparison

CONCLUSIONS AND DELIVERABLES

- Three new designs furbished
 - 29% max reduction in mass (Design 3)
 - 34% max reduction in stress (Design 1)
- Three material candidates were proposed
 - Al 7068 was safest option overall
 - Mg AZ61A has highest potential for cost and mass savings with a reduction in strength
- Process developed
 - Data collection and analysis
 - Finite Element Analysis
 - Pinpointing stress & redesign

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Aluminum Suppliers)

32 6e+01
049e+01
772e+01
494e+01
217e+01
940e+01
663e+01
386e+01
109e+01
3 15e+ 00
543e+00
772 e +00
678e-07

Questions?

(Strain rate effects on tensile deformation of 2024-0 and 7075-0 aluminum alloy sheet)

Figure 2 The relationship between DIF and common logarithmic of strain-rate for 7075 aluminum

		Axis Maximum and Minimum Acceleration Values Live Fire Data																							
				Front	of Rail		Rear of Rail						Front Scope Position							Rear Scope Position					
Shooter Size	Shot Number	X _{max}	X _{min}	Y _{max}	Y _{min}	Z _{max}	Z _{min}	X _{max}	X _{min}	Y _{max}	Y _{min}	Z _{max}	Z _{min}	X _{max}	X _{min}	Y _{max}	Y _{min}	Z _{max}	Z _{min}	X _{max}	X _{min}	Y _{max}	Y _{min}	Zmo	
	1	172.3	-140.7	144.8	-112.9	123.6	-101.0	111.4	-104.1	110.1	-85.9	73.6	-62.5	194.4	-148.0	156.0	-143.7	174.4	-145.5	144.6	-136.6	150.6	-126.0) 121	
	2	161.6	-124.5	129.3	-121.5	98.4	-87.6	110.9	-105.4	96.0	-125.5	79.2	-76.1	145.3	-138.4	142.9	-124.6	146.9	-118.1	148.4	-137.6	156.1	-126.8	; 12 3	
Large	3	131.0	-144.6	182.2	-136.8	121.4	-112.0	121.6	-89.7	111.1	-104.6	75.5	-65.6	183.6	-135.5	163.4	-124.4	168.6	-127.5	209.2	-149.2	204.9	-148.4	143	
	4	144.8	-137.6	100.5	-107.9	98.4	-94.2	113.3	-98.6	111.8	-105.1	129.5	-81.7	132.9	-137.9	160.6	-141.6	183.0	-124.2	144.0	-132.3	149.9	-128.4	122	
	5	157.5	-138.7	129.1	-109.3	83.5	-89.2	122.4	-113.4	89.1	-106.5	152.8	-76.3	155.5	-143.1	167.7	-139.2	146.2	-110.5	192.3	-153.5	190.5	-154.9	133	
Medium	1	132.4	-126.7	90.2	-91.8	105.8	-98.5	199.5	-144.7	122.1	-126.1	97.5	-90.3	137.5	-143.8	131.4	-132.1	130.2	-127.1	126.5	-118.4	150.7	-127.3	i 14 3	
	2	125.9	-83.0	73.1	-103.0	85.9	-72.4	98.6	-121.0	97.7	-108.3	87.4	-78.1	186.0	-130.5	184.6	-132.2	187.9	-119.8	127.6	-146.8	178.7	-127.6	i 131	
	3	113.6	-102.8	87.2	-105.2	67.3	-77.3	72.7	-65.2	75.5	-74.7	56.7	-71.8	164.0	-152.9	181.9	-143.7	174.9	-119.6	138.1	-141.0	152.9	-123.8	3 119	
	4	122.3	-107.3	139.8	-127.0	92.2	-85.0	85.2	-76.1	69.4	-66.2	58.2	-65.6	192.7	-133.8	139.1	-129.3	166.7	-139.2	118.6	-131.4	176.8	-152.9	100	
	5	111.2	-129.3	135.5	-107.9	92.2	-98.3	106.3	-111.6	104.5	-110.5	62.8	-91.7	162.6	-139.8	148.3	-127.6	136.9	-119.1	172.1	-137.4	176.9	-150.5	i 130	
	1	98.5	-91.4	65.6	-62.2	81.8	-59.1	60.2	-83.3	67.5	-73.8	54.8	-54.0	135.1	-133.3	151.5	-129.6	124.0	-123.9	132.0	-124.9	125.9	-114.9	140	
	2	88.8	-73.4	100.9	-67.5	73.9	-87.5	61.6	-91.3	60.7	-77.1	62.9	-62.9	172.4	-132.9	145.2	-139.2	150.1	-143.4	174.7	-144.6	127.2	-135.5	i 97	
Small	3	79.4	-86.4	76.5	-79.3	90.8	-61.1	78.8	-124.2	111.1	-102.7	81.1	-57.7	172.4	-132.9	145.2	-139.2	150.1	-143.4	175.0	-131.0	178.7	-104.4	+ 122	
	4	91.7	-86.0	69.3	-67.0	60.9	-54.8	61.7	-117.7	78.4	-89.2	70.0	-55.4	205.5	-141.5	199.3	-144.0	139.4	-114.9	183.9	-150.9	164.6	-140.9	139	
	5	112.0	-88.3	89.1	-96.9	84.1	-65.4	128.5	-122.8	110.4	-129.3	58.3	-72.8	145.8	-150.7	207.6	-143.9	142.8	-112.2	145.6	-139.3	182.2	-141.1	. 123	
	1	120.5	-118.2	107.1	-98.4	80.8	-70.1	142.1	-124.4	112.2	-143.2	106.6	-95.0	152.6	-143.0	213.4	-152.2	162.3	-131.7	134.5	-117.2	148.6	-133.6	i 139	
	2	128.3	-136.0	116.6	-117.7	100.0	-85.7	164.7	-125.0	130.1	-130.8	104.4	-92.1	180.9	-157.3	178.2	-155.3	154.8	-149.5	169.4	-132.3	106.7	-139.0	131	
Unsupported	3	161.4	-117.1	96.7	-131.6	92.7	-96.6	188.6	-151.9	130.8	-130.5	146.0	-132.5	135.1	-136.1	173.0	-145.9	131.8	-111.0	150.6	-140.3	178.0	-124.6	i 106	
	4	135.6	-145.2	100.9	-89.6	106.0	-94.3	165.0	-159.6	172.1	-145.2	141.6	-116.4	173.1	-150.2	192.1	-145.8	171.1	-137.3	141.8	-122.1	138.0	-129.2	110	
	5	123.1	-141.1	126.7	-88.6	115.7	-105.3	188.5	-126.7	153.9	-125.0	122.0	-127.6	160.4	-146.7	189.1	-149.8	155.0	-132.4	110.0	-130.4	114.6	-126.2	113	

